



EDL Engineering Constraints

Adam Steltzner
Mike Watkins



Overview



- Landing site safety assessment involves evaluation of broad spectrum of risks
 - Uncertainty exists in data set used to evaluate
 - Mars is frequently unkind
 - Risk evaluation involves judgment
 - “Less” risk is always better than “more”
- EDL engineering safety constraints is a discretized set of thresholds
 - Environmental parameters above (beyond?) which additional risk/work exists
 - Discretization process injects error
 - Discretizing a continuum of environmental characteristic into a finite set
 - We only know to be afraid of that which we can conceive of
 - The engineering safety constraints are complex
 - Some of the constraints are very firm and more brittle than others when exceeded
 - Some of the constraints are not as firm
 - Constraints are interconnected
- We offer a brief primer on a subset of the constraints
 - Supports the search for other things to be afraid of
 - Informs science community of environmental characteristic to be considered in the site evaluation
 - General understanding
 - Time constrains discussion



Landing Site Elevation



Landing Site Workshop 2

Mars Science Laboratory

Constraint

- Requirement: $\leq +1$ km MOLA
- Safe haven:
 - ≤ -1 km MOLA
 - ≤ -2 km MOLA most desirable

Possible Trades

- Terrain tolerance
 - Lower elevation sites yield timeline and propellant margin that can be spent on terrain relief tolerance
 - Example: -1 km MOLA site can yield >50 m additional terrain relief tolerance
- Precision
 - Lower elevation sites can sometimes be used to increase precision performance
 - 0.5 km decrease in site elevation may yield ~ 0.5 km precision improvement
- Robustness
 - Lower elevation sites yield timeline and propellant margin allowing increased system/environment uncertainty

Why We Care

- High elevation sites require the vehicle to be decelerated faster
 - Need enough time to complete EDL events required for safe landing
- More fuel is required to land at high elevation sites
 - Parachute under-utilized (not enough time)
 - Atmospheric density is lower
- Low elevation sites allow the radar to better “see” the eventual landing site

Criticality/Firmness of Constraint

- Cannot exceed $+1$ km elevation
 - System has extremely limited timeline and propellant margin at $+1$ km sites
 - Elevation capability increase unlikely
- Elevation directly affects critical resources
 - Altitude/timeline margin
 - Propellant margin
- Timeline/fuel risks are greatly reduced at or below -0.5 km MOLA



Landing Site Terrain



Landing Site Workshop 2

Mars Science Laboratory

Constraint

- Requirement:
 - $\leq 20^\circ$ slope for 2 – 10 km length scales
 - Also applies to warning track
 - ≤ 43 m relief at 0.2 – 1.0 km length scales; increasing linearly to 720 m at 2 km length scales
 - $\leq 15^\circ$ slope for 2 – 5 m length scales

Possible Trades

- Site elevation
 - Reduced site elevation generates propellant and altitude margin that could be used for additional 0.2 – 1.0 km length scale terrain relief (reverse also applies)
 - Example: 1 km lower site yields ~50 m additional terrain relief tolerance
- Precision
 - Lower elevation sites can sometimes be used to increase precision performance
 - Increased precision performance may shrink the area over which the terrain restrictions apply (smaller warning track, smaller landing area)
- Winds
 - Reduced horizontal wind magnitudes and wind uncertainties may shrink the area over which the terrain restrictions apply (smaller warning track, smaller landing area)
- Robustness
 - Reduced terrain relief yields propellant margin allowing increase system/environment uncertainty

Why We Care

- 2 – 10 km length scales: large slopes may spoof the system into beginning powered flight too high or too low
 - Why a warning track: if we land near the edge of the landing ellipse, the radar will be looking at terrain outside the ellipse
- 43 m relief at 0.2 – 1.0 km length scales: a certain amount of propellant and altitude margin is allocated to terrain relief tolerance in powered descent
- 1 km to 2 km length scales: transition smoothly between two length scale restrictions above
- 2 – 5 m length scales: ensures stability and trafficability of the rover in touchdown conditions

Criticality/Firmness of Constraint

- All terrain relief/slope constraints appear tradable, especially at lower altitude sites
- Some statistical element to terrain constraints
 - Where terrain features are located in ellipse (not all locations in ellipse are equally likely)
- Consequence of terrain relief has different criticality depending on direction
 - Down-slopes cause additional propellant usage
 - Up-slopes cause reduced altitude/timeline margin

Pre-decisional draft: for planning and discussion purposes only.

The data/information contained herein has been reviewed and approved for release by JPL Export Administration on the basis that this document contains no export-controlled information.



Day of Landing Winds



Landing Site Workshop 2

Mars Science Laboratory

Constraints

- Requirement:
 - ≤ 25 m/s horizontal uncertainty between 6.5 – 20 km altitude MOLA
 - ≤ 20 m/s horizontal uncertainty between 3 – 6.5 km altitude MOLA
 - ≤ 20 m/s downward vertical magnitude between 1 – 5 km altitude above ground level

Possible Trades

- Site elevation
 - Lower site elevation yields altitude/timeline margin that can be spent on horizontal wind uncertainty
 - Example: ~300 – 400 m lower landing site elevation can offset ~10 m/s additional horizontal wind uncertainty
- Precision
 - Lower horizontal wind uncertainties increase precision
 - Example: 10 m/s lower horizontal wind uncertainty may reduce landing error ~0.4 – 0.8 km
- Terrain tolerance
 - Lower downward vertical wind magnitudes yield propellant margin that can be spent on terrain relief tolerance
 - Example: 10 m/s lower downward wind can yield ~50 m additional terrain relief tolerance

Why We Care

- 6.5 – 20 km MOLA horizontal wind uncertainty affects altitude and precision performance
 - Spreads parachute deploy Mach “error”
- 3 – 6.5 km AGL horizontal wind uncertainty affects altitude performance
 - Spreads heatshield separation Mach “error”
- 1 – 5 km AGL vertical wind magnitude affects propellant margin
 - Increases powered flight starting velocity and altitude

Criticality/Firmness of Constraints

- Wind constraints are not as firm as other constraints (e.g. site elevation)
- Wind constraints may be exceeded if traded for reduction in other constraints
- Site elevation is most valuable trading chip
 - Impacts altitude/timeline and propellant critical resources



Landing Precision (Miss Distance)



Landing Site Workshop 2

Mars Science Laboratory

Constraint

- Requirement: ≤ 12.5 km in downtrack direction; $\leq \sim 10$ km in crosstrack direction
- Safe haven: ≤ 16 km in downtrack direction

Possible Trades

- **Not really tradable**
- Atmosphere
 - Quiescent atmosphere conditions (wind, density structures, etc.) will improve precision performance

Why We Care

- Constraint defines expected landing precision capability of vehicle across range of landing sites
- Terrain safety constraints apply within the potential landing ellipses
 - Exceptions: “warning track” constraint, some atmosphere constraints

Criticality/Firmness of Constraint

- Unlikely to be able to significantly improve performance across landing sites
- Some capability variability with latitude
 - Orientation of ellipse
 - Precision performance
- Some statistical element to precision capability
 - Not all portions of the landing ellipse are equally likely
 - Ellipse more circular than past missions
- Ellipse performance will “never” be known
 - Largely driven by day-of-landing Mars atmospheric conditions



Landing Site Rocks Distribution



Landing Site Workshop 2

Mars Science Laboratory

Constraint

- Requirement: probability that a rock higher than 0.55 m occurs in a random sampled area of 4 m² should be less than 0.5%

Possible Trades

- **Not really tradable**
- Robustness
 - Reductions in likelihoods of other failure mechanisms may enable a larger failure likelihood allocation for rocks

Why We Care

- Rover mobility system can accommodate rocks up to 0.55 m before the rover lower structure is damaged
- 0.5% of the project allocation for possible failure has been assigned

Criticality/Firmness of Constraint

- Risk of belly pan strike directly impacts probability of safe landing
 - No other constraints can be traded to reduce this impact
 - Vehicle cannot be modified to increase clearance
- Some statistical element to rock constraint
 - Where rocks are located in ellipse (not all locations in ellipse are equally likely)
 - Percentage of chance of failure that can be allocated



Detailed Site Assessment Process



Landing Site Workshop 2

Mars Science Laboratory

- Site assessment involves detailed environmental characterization and design tuning
 - Monte Carlo performance simulation used as assessment tool
- Environmental characterization process
 - Terrain digital elevation maps required for several purposes
 - Radar/terrain interaction model and assessment of relief/slope tolerance require ~10 m resolution DEMs
 - Touchdown interaction model and rock strike probability assessment likely require high resolution DEMs
 - Data sets: CTX, HiRISE, photoclinometry
 - Computationally intensive detailed atmosphere modeling required to assess altitude and precision capability at each site
 - Mesoscale modeling captures site specific atmosphere features for guided entry
 - LES modeling may be required to capture low altitude features that impact performance
- Design tuning at each site provides most realistic day of landing risk assessment
 - Options exist for modifying the way the vehicle is flown for each site
 - Tuning guided by performance simulation results
 - Exploring design options is time and personnel intensive
- Team has laid groundwork for site assessments for ~5 sites
 - Throughput is largely team constrained
 - Results generation is labor intensive
 - Results synthesis is a small group affair
 - Radar models (utilizing DEMs) in development
 - Atmosphere modeling community engaged
 - Evaluation process has been preliminarily exercised



EDL Constraints Summary



Landing Site Workshop 2

Mars Science Laboratory

Constraint Summary:

- Landing site elevation is a valuable trading chip
 - System taxed along many dimensions at high altitude sites
 - Margin at and above +1 km elevation sites is very small
 - Winds and terrain tolerance may be purchased with site elevation reductions
- Landing Precision
 - Performance may vary across launch/arrival/latitude/altitude space
 - Sites with uncertain atmospheric conditions (winds and density) degrade performance
- Rocks abundance above 0.55 m directly impacts landing safety
 - Reductions in other constraints do not directly yield additional capability

EDL Safety Assessment Posture:

- Detailed site assessment is needed to understand true integrated risk at any one site
 - Data sets for evaluation are still pending
- Meeting the engineering constraints does not make all sites equal
 - Safety assessment values safer sites more
 - Safety assessment requires relative comparisons of detailed site assessments